

APPROVAL CERTIFICATE

EASA.21J.055

Pursuant to Regulations (EU) 2018/1139 and (EU) 748/2012 and subject to the conditions specified below, the Agency hereby certifies

SABENA TECHNICS BOD
19, Rue MARCEL ISSARTIER
33700 Merignac
France

as a DESIGN ORGANISATION

approved according to Part 21, Section A, Subpart J.

CONDITIONS :

1. The approval is limited to that specified in the enclosed Terms of Approval, and
2. This approval requires compliance with the procedures specified in the Design Organisation Handbook, reference DI/T 2001/500/PL, in the latest revision, and
3. This approval is valid whilst the approved Design Organisation remains in compliance with Part 21, Section A, Subpart J.
4. Subject to compliance with the foregoing conditions, this approval shall remain valid until surrendered or revoked.

For the **European Union Aviation Safety Agency**,

Date of issue: 09/11/2020



Hans LUNDSTRÖM
Senior DOA Team Leader

Terms of Approval

Design Organisation Approval Certificate

EASA.21J.055

1 Scope

This Design Organisation Approval is applicable for the scope defined in Annex A for design work with regard to the airworthiness, operational suitability and environmental characteristics of the products.

2 Privileges

- a) (Reserved)
- b) (Reserved)
- c) The holder of this design organisation approval shall be entitled, within the scope of this terms of approval, and under the relevant procedures of the design management system:
 1. to classify changes to a type certificate or to a supplemental type certificate and repair designs as “major” or “minor”;
 2. to approve minor changes to a type certificate or to a supplemental type certificate and minor repair designs under this Annex (Part 21) or under Annex Ib (Part 21 Light);
 3. [not applicable];
 4. [not applicable];
 5. [not applicable];
 6. to approve for certain aircraft the flight conditions under which a permit to fly can be issued in accordance with point 21.A.710(a)(2), except for permits to fly to be issued for the purpose of point 21.A.701(a)(15);
 7. to issue a permit to fly in accordance with point 21.A.711(b) for an aircraft it has designed or modified, or for which it has approved, in accordance with point 21.A.263(c)(6), the flight conditions under which the permit to fly can be issued, and where the holder of this design organisation approval itself:
 - (i) controls the configuration of the aircraft, and
 - (ii) attests conformity with the design conditions approved for the flight;
 8. [not applicable];
 9. [not applicable].

3 Obligations

The holder of this design organisation approval shall, within the scope of this terms of approval:

- a) maintain the handbook required under point 21.A.243 in conformity with the design management system;
- b) ensure that this handbook or the relevant procedures included by cross-reference are used as a basic working document within the organisation;
- c) determine that the design of the products, or of the changes or repairs to them, complies with the applicable type-certification basis, technical specifications concerning the making of declarations, operational suitability data certification basis, and the environmental protection requirements and have no unsafe features;
- d) provide the Agency with statements and associated documentation confirming compliance with point (c), except for approval processes carried out in accordance with point 21.A.263(c);
- e) provide to the Agency data and information related to the actions required under point 21.A.3B;
- f) determine, in accordance with point 21.A.263(c)(6), the flight conditions under which a permit to fly can be issued;
- g) establish, in accordance with point 21.A.263(c)(7), compliance with points (b) and (e) of point 21.A.711 before issuing a permit to fly to an aircraft;
- h) designate data and information issued under the authority of the approved design organisation within the scope of this terms of approval with the following statement: "The technical content of this document is approved under the authority of the DOA ref. EASA.21J.055".
- i) comply with Part 21, Section A, Subpart A.

Date of issue: 18/02/2025


CIRIO PIRONE
Senior DOA Team Leader

Annex A

Scope of work

	TC	STC	major changes	minor changes	major repairs	minor repairs	flight conditions	permit to fly
Large and small aeroplane								
Avionics								
All areas		■	■	■	■	■	■	■
Cabin								
All areas		■	■	■	■	■	■	■
Electrical Systems								
All areas		■	■	■	■	■	■	■
Environmental Control Systems								
All areas		■	■	■	■	■	■	■
Structures								
All areas		■	■	■	■	■	■	■
Large rotorcraft								
Avionics								
Communication systems			■	■	■	■		
Navigation systems			■	■	■	■		
Recording systems			■	■	■	■		

Legend:

■	Title for category of product
■	Title for design scope
■	Title for design area

■	Within scope
□	Outside scope

List of products

Not applicable

Limitations

Limitations common to all products and activities

1. Development of Operational Suitability Data excludes the OSD constituents FCD, SIMD and MCSD.
2. The privilege under paragraph 2(c)(2) excludes changes to the aircraft flight manual and supplements other than documentary changes.

Product	Limitations particular to each product
Large rotorcraft	For non-TCH activity: limited to changes and repairs to own STC in order to assure continuous airworthiness